

Facility for Satellite Formation Flying Control System Verification and Electric Propulsion Systems

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Supported by EPSRC, EADS Astrium, Aero-Astro

Outline of talk

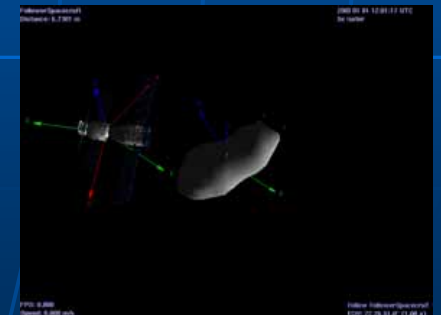
- Modes of operation
- The GSF Testing
- Metrology Infrastructure
- Satellite Frames available
- Future development and use of facility
- Electric propulsion capabilities



Two modes of operation

A two phase approach is taken to test a satellite formation flying system:

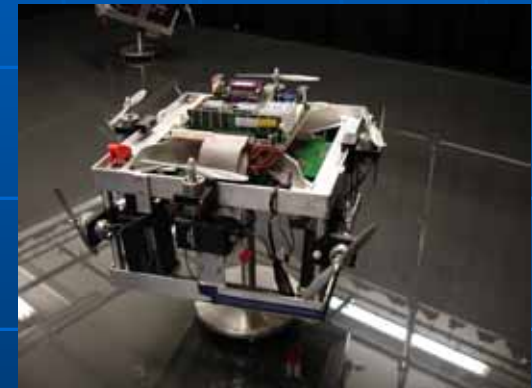
- *Testing of the processors and communication electronics* with a limited number of sensors and actuators (gyros, reaction wheels and air-jets) This is called ground-based satellite frames testing (GSFT).
- *Testing of the actual control algorithms*, as they would be used in space, using hardware in the loop simulations: control electronics are real but the sensor, actuator and physical dynamics are simulated. This is called hardware in the loop testing (HILT).



The GSF Testing

GSFT (= *ground-based satellite frames testing*) allows testing of systems software and hardware in operation with real components:

- solid state gyros (actual)
- accelerometers (actual)
- reaction wheels (actual)
- on board computers (actual)
- communication electronics (actual)
- propulsion substitutes (similar in dynamics)



Testing of satellite autonomous cooperation

Frames in preparation for testing



Metrology infrastructure



1/26/2007

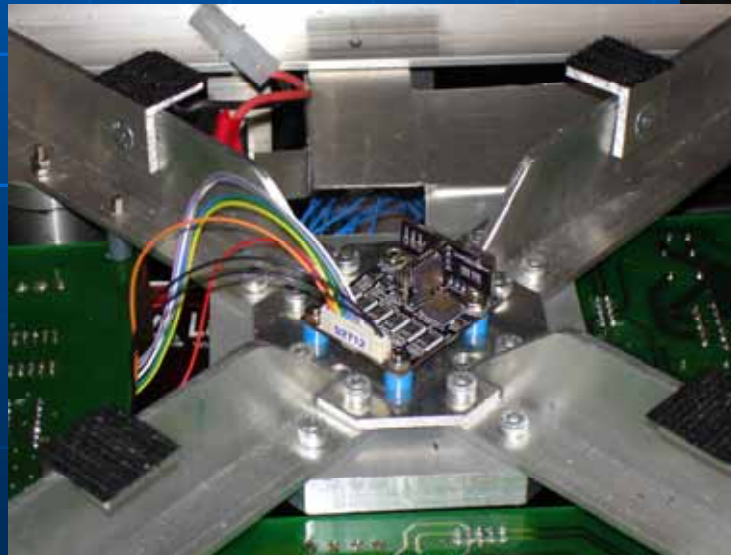
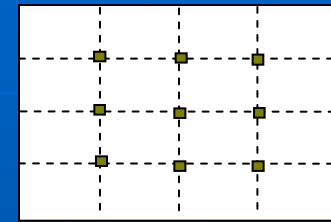
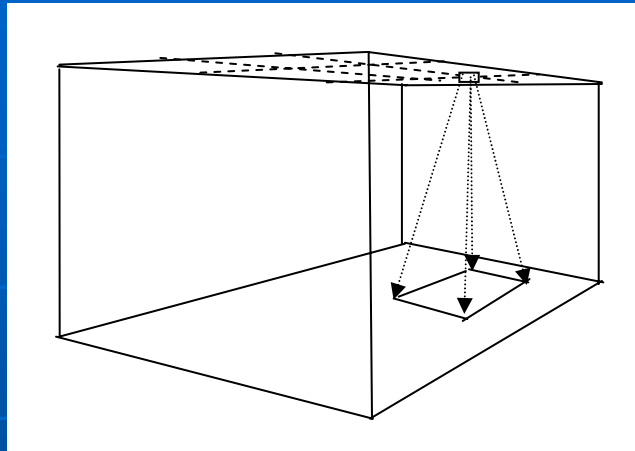
PPARC - ESTEC

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Metrology features

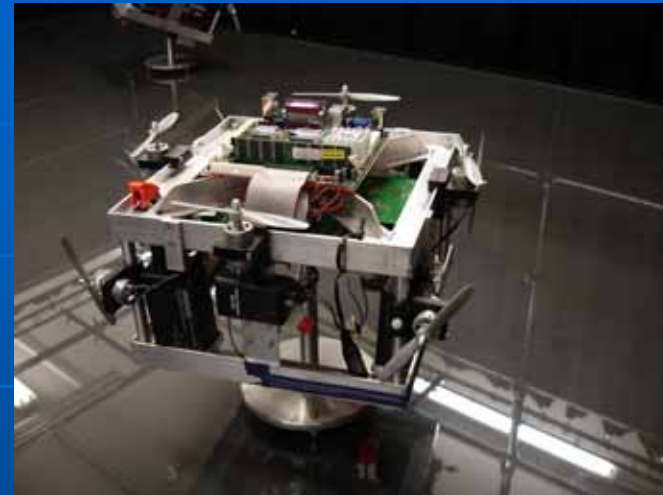
A calibrated **9 camera based observation** system which can monitor the positions and attitudes of each frame in realtime. The **highly sensitive solid state gyros** and the camera based position and attitude estimates are processed through a data fusion system which provides **realtime estimates of the 12 dimensional state** of each frame moving on the table.



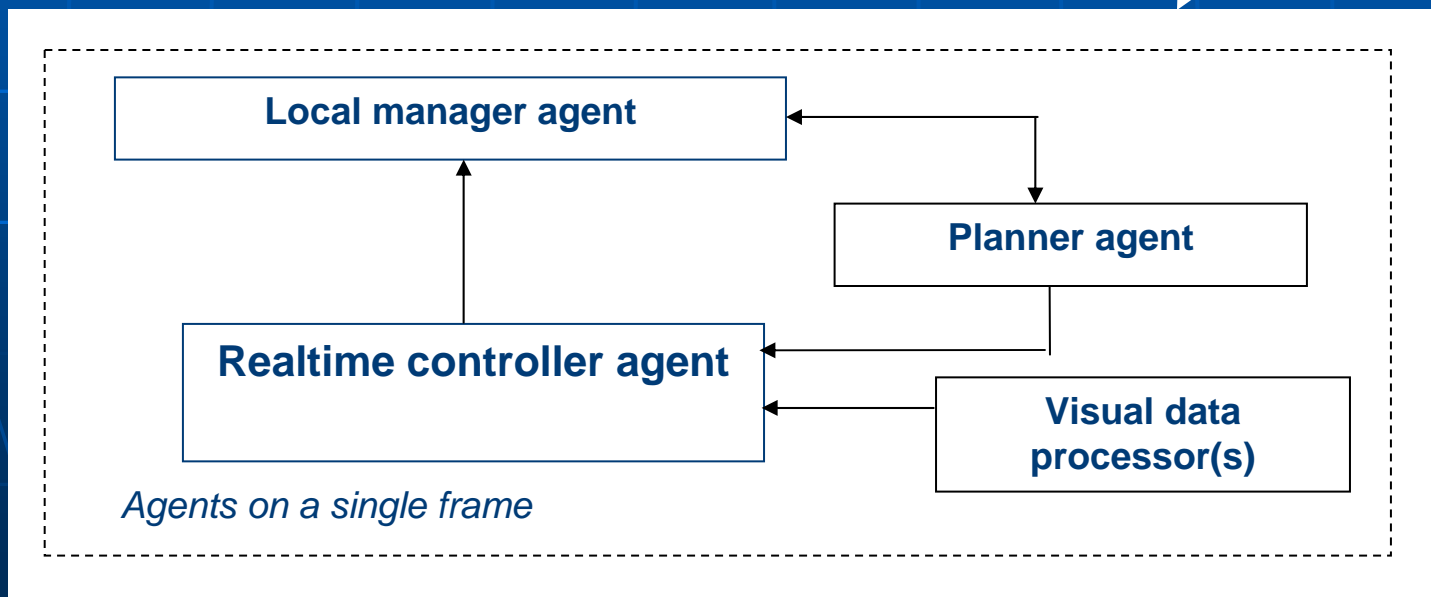
1 cm
accuracy for
coarse
formation
flying

Available GSFT frames

- 3 with model aircraft propeller actuation (8 bidirectional actuators in each frame)
- 3 frames with air-jet actuators (16 unidirectional actuators on each frame)
- and 3 frames for future customer use



Anticipated use: testing of multi-agent systems on hardware



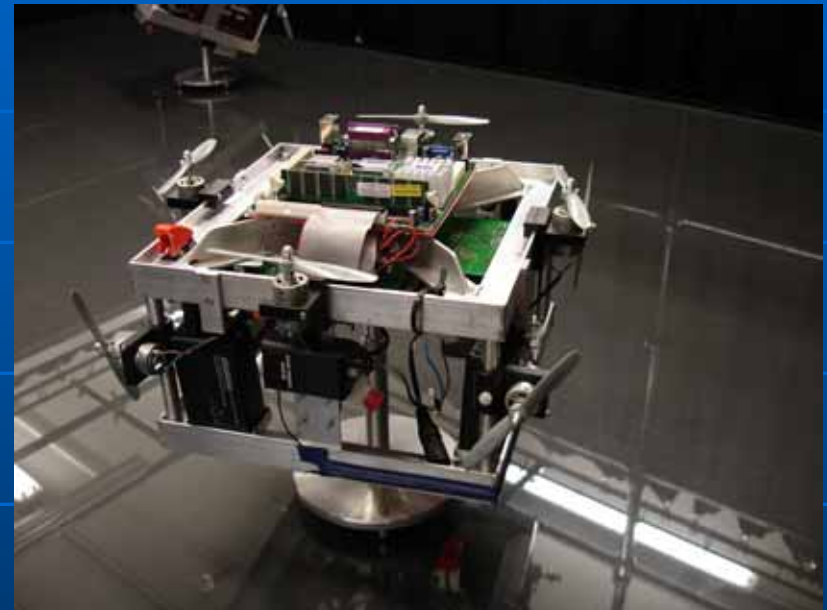
Future developments

- Precision positioning using laser range finders
- System verification software added
- Compliance with software standards
- Ready made modules with hardware standards
- Etc



Conclusions on SFF facility

- Facility almost ready for industrial use
- GSFT and HILT are complementary – the two together address all important issues before space use
- High quality design and built – cheaper to use our facility then recreate something similar for science missions



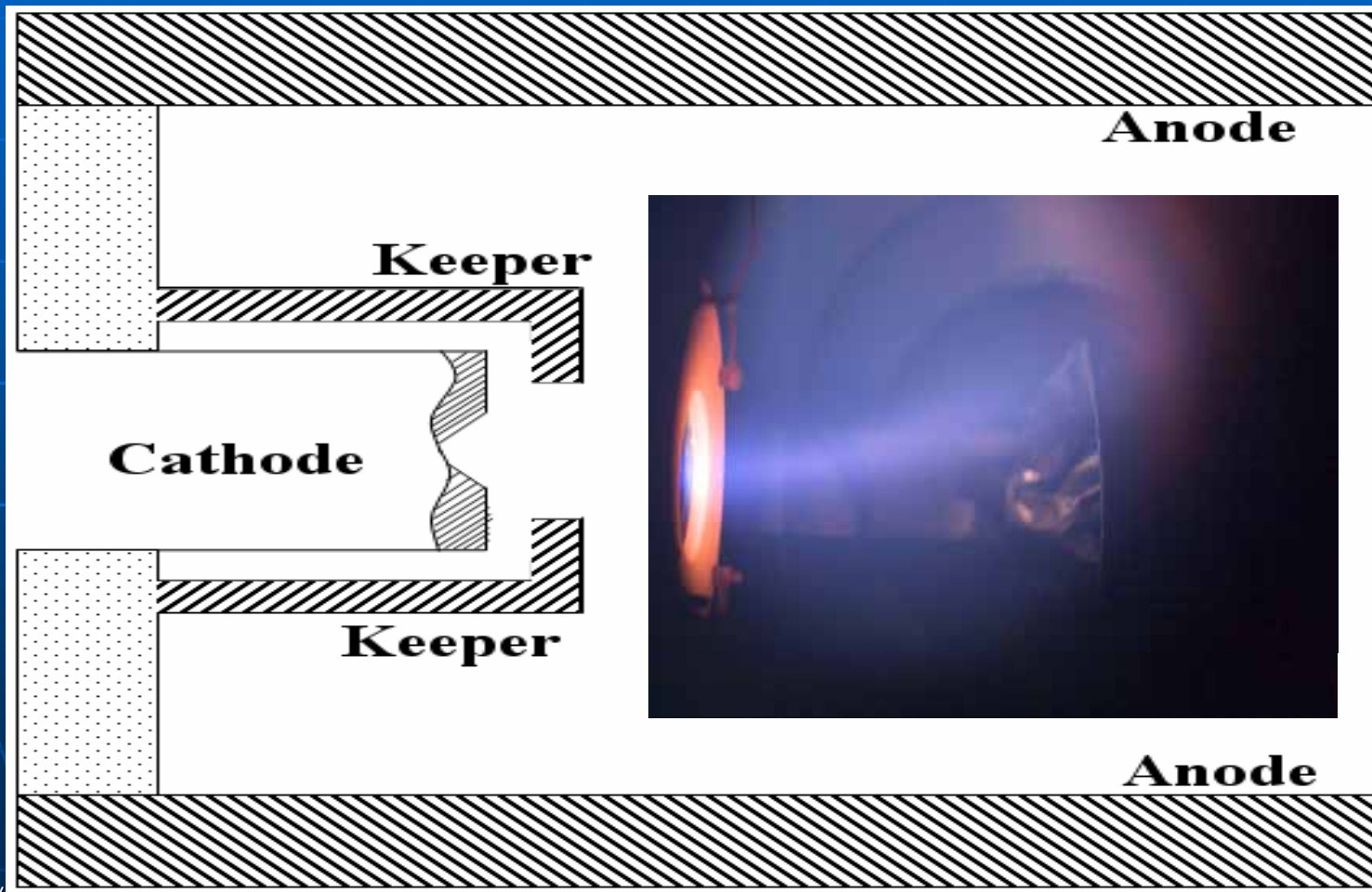
THANK YOU !

Hollow cathode micro-thrusters(HCMTs)

- Based on well developed and flight – qualified T series hollow cathodes
- Thrust range – 1-5 mN
- Specific Impulse in 100-500s range
- Possible application to 'all-electric' spacecraft : gridded ion thrusters for primary propulsion + HCMTs for attitude control → being studied for possible application to European Student Moon Orbiter(ESMO)



Preliminary Design of a HCMT



New Type of mini- ion thruster

- QinetiQ concept (patent pending, Neil Wallace)
- ESTEC contract : Southampton sub-contract to work on discharge chamber
- Applications to fine control for future formation flying missions (eg Lisa, Darwin, Xeus)